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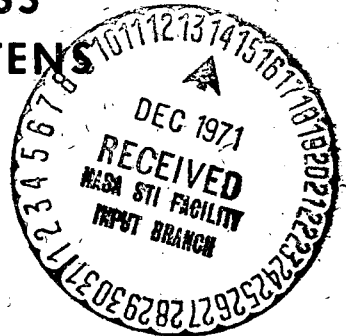
MAGNETIC FIELD MEASUREMENTS BY PIONEER 6

I. Hourly Averages of the Field
Elements from 17 December 1965
to 5 September 1967 (Bartels' Solar
Rotation 1811 to 1834)

NORMAN F. NESS
FRANKLIN W. OTTENS

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**GODDARD SPACE FLIGHT CENTER
GREENBELT, MARYLAND**

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MAGNETIC FIELD MEASUREMENTS BY PIONEER 6

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Norman F. Ness
Franklin W. Ottens

Laboratory for Extraterrestrial Physics
NASA-Goddard Space Flight Center
Greenbelt, Maryland USA 20771

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1. Introduction

The Pioneer 6 spacecraft is a spin stabilized interplanetary probe launched from the Eastern Test Range, Cape Kennedy, Florida on 16 December 1965. The geocentric orbit of the spacecraft has an aphelion of 0.987 AU, a perihelion 0.814 AU, inclination to the ecliptic of 0.2° and orbital period of 311 days. This spacecraft was the first of a series of four Pioneer spacecraft with identical structures but slightly different complements of experiments. A NASA-GSFC magnetic field experiment was proposed, accepted and flown on Pioneers 6 and 7 (in 1966). Subsequently, unique circumstances presented an opportunity for flight of a similar experiment by the GSFC group in collaboration with the University of Rome, Laboratory for Space Plasma Research group on Pioneer 8 (in 1967).

The magnetic field experiment detector is a monoaxial fluxgate magnetometer sensor mounted on the end of a boom at a distance of 2.1 meters from the spacecraft spin axis. The initial spin period was approximately one second and the field measurements were synchronized with the rotation of the spacecraft. The monoaxial sensor was oriented at an angle of $54^{\circ} 45'$ to the spin axis so that during one spacecraft rotation three samples of the magnetic field at equal intervals yielded three independent measurements in mutually orthogonal directions. These basic measurements, obtained in a time interval of $2/3$ of a second, define the total vector magnetic field in a spacecraft coordinate system.

The spin axis of the spacecraft was oriented orthogonal to the ecliptic plane with the high gain antenna of the spacecraft pointing to the south ecliptic pole.

Data transmission from the spacecraft experiment was synchronized with telemetry, which ranged from 8 to 512 bits/sec during the lifetime of the spacecraft. A complete three component measurement was not read out during each spin period and so individual data points are not equally spaced in time (see Searce et al., 1968 for description of the experiment).

The Pioneer 6 instrument was a single range magnetometer ($\pm 64\gamma$) which with the 8 bit quantization of the analog to digital converter yielded a digital window size of $\pm 0.25\gamma$. The RMS noise level of the sensor was 0.12γ and the bandwidth 0 to 5 cps. The magnetic field of the spacecraft and associated solar array as measured at the sensor location was estimated to be less than 0.3γ . Periodic sensitivity calibrations were made once each day as well as zero level checks approximately once every several months. Pioneer 6 carried the first device to check the zero-level of a fluxgate magnetometer sensor inflight. The projection of the orbit described by Pioneer 6 on the ecliptic plane relative, to the earth-sun-line, is shown in Figure 1.

2. Data

The magnetic field elements have been computed in a spacecraft centered solar ecliptic system in which the X axis points from the spacecraft to the sun, the Z axis to the ecliptic north pole and the Y axis being the third axis of the orthogonal right handed coordinate system.

The standard data analysis was performed to yield 30 second averages of the field components and the field magnitudes as

$$\bar{X} = \frac{1}{n} \sum_{i=1}^n X_i \text{ etc.}, \bar{F} = \frac{1}{n} \sum_{i=1}^n F_i, \bar{F} = \sqrt{(\bar{X})^2 + (\bar{Y})^2 + (\bar{Z})^2} \quad (1)$$

Also RMS deviations $(\delta_X, \delta_Y, \delta_Z)$ of the magnetic field were computed routinely for each component and for the field magnitude. The Pythagorean variance, which is an invariant of the coordinate system, is obtained from the component deviations as

$$\delta = \sqrt{\delta_X^2 + \delta_Y^2 + \delta_Z^2} \quad (2)$$

The choice of a 30 second time interval was based upon a desire to provide correlative data to each of the plasma probes on the Pioneer 6 and 7 spacecraft whose fastest cycle times were on the order of one minute.

Data shown in Figures 2 through 22 represent for each solar rotation the hourly averages of the field elements \bar{F} , \bar{F} , θ (latitude) and ϕ (longitude). The data are plotted in 27 day increments corresponding to Bartels' solar

rotation numbering system. The two variances D_1 and D_2 are obtained by different means. D_1 is a variance computed from equation 2 for each hourly interval using the hourly averaged components (and deviations from them). D_2 is the hourly average of the individual Pythagorean variance for each 30 time interval. The days of the year are indicated as calendar days and most of the missing data corresponds to gaps in tracking coverage of the spacecraft. Publications utilizing Pioneer 6 data which have already appeared are given in the reference list. This does not include those publications by other non-GSFC associated authors who have used Pioneer 6 data.

3. Summary

The most important scientific discoveries of the Pioneer 6 magnetic field experiment have related to the microstructure of the interplanetary medium as well as the large scale structure of the interplanetary magnetic field. In addition, correlative studies with the MIT plasma probe have provided further insight into the nature of the microstructure of the interplanetary plasma.

The 30 second average detail data and hourly average data are available from the National Space Science Data Center at NASA-GSFC (Mail Code 601), Greenbelt, Maryland 20771. This present document provides, in summary graphical form, a working data set which was used internally at the NASA-GSFC and parts of which have been made available to other groups.

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List of Figures

- 1 Projection of the orbit of Pioneer 6 on the ecliptic plane. Position of the spacecraft for the first day of each month is indicated by month/year.
- 2- Hourly averages of the magnetic field elements F , F , θ , ϕ , $D1$ and
22 $D2$ for the Bartels' solar rotation number specified on the left hand side. The lack of two values for F , F , means that the spacecraft was in the DCS (Duty Cycle Storage) mode of operation with delayed transmission of data. The character A represents the "folding over" of the corresponding scale by the maximum value of the scale as shown on the left, Hence the correct value is to be read = A + Max Scale. Gaps in data are due to lack of telemetry recording by NASA-JPL DSN stations.

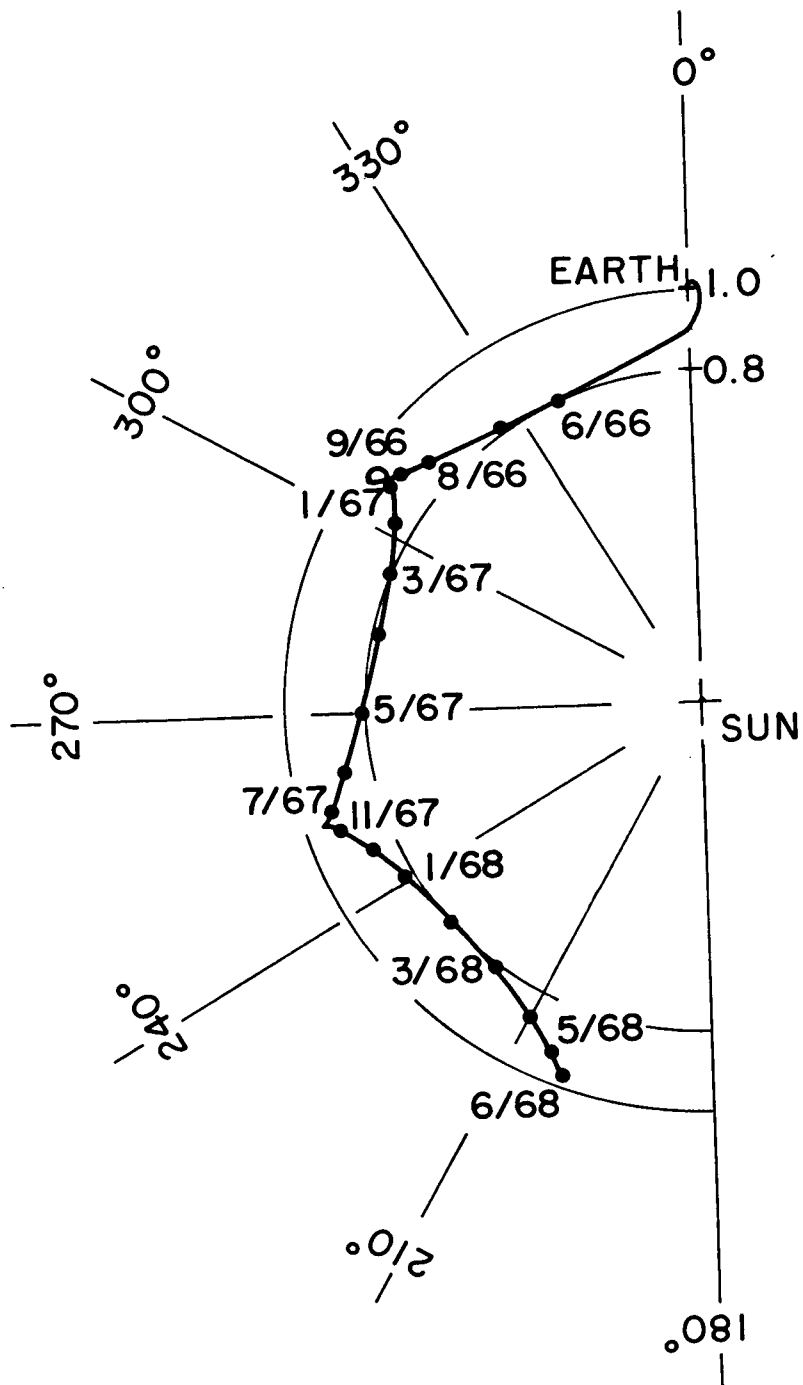


FIGURE 1

SOLAR ROTATION 1811

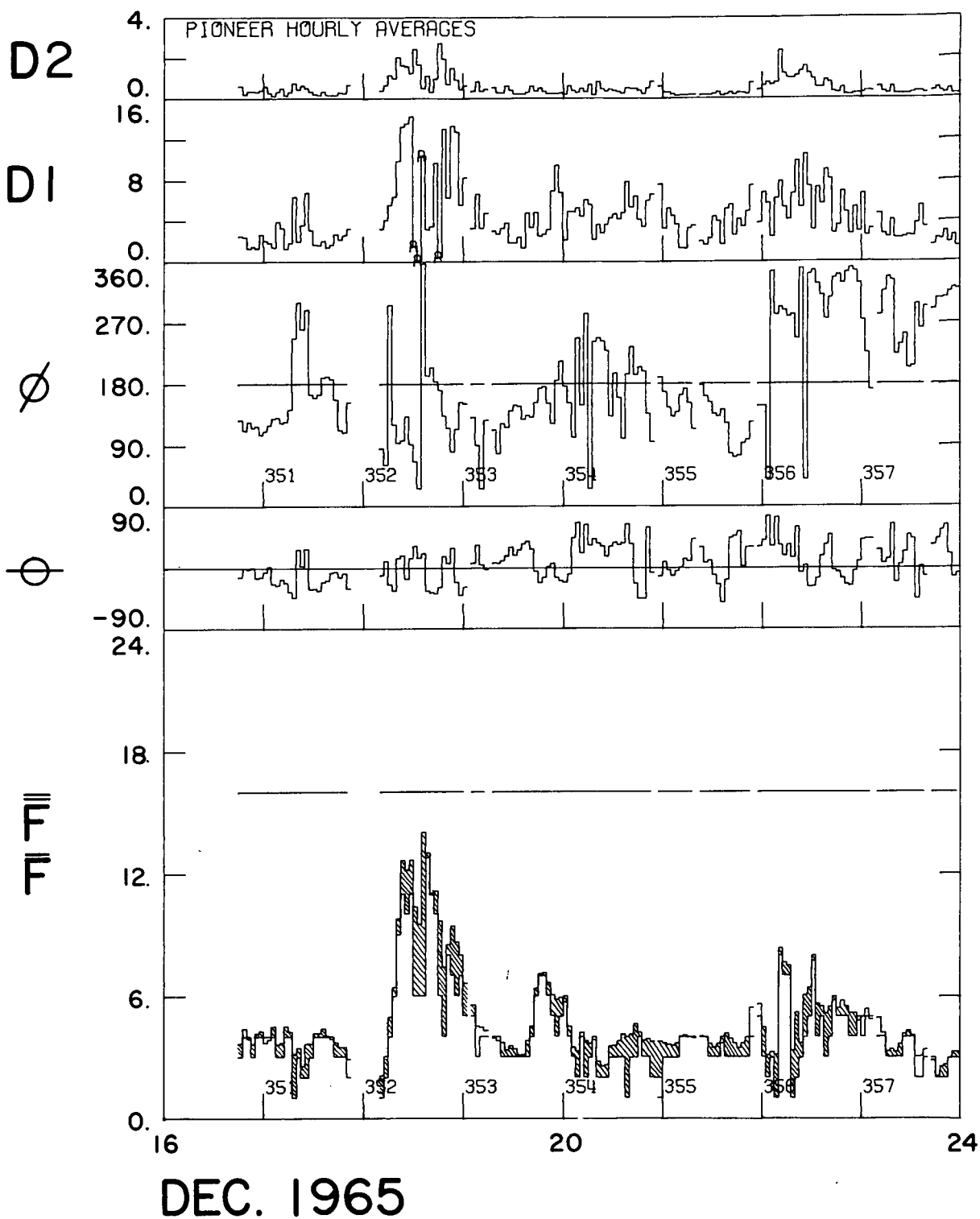


FIGURE 2

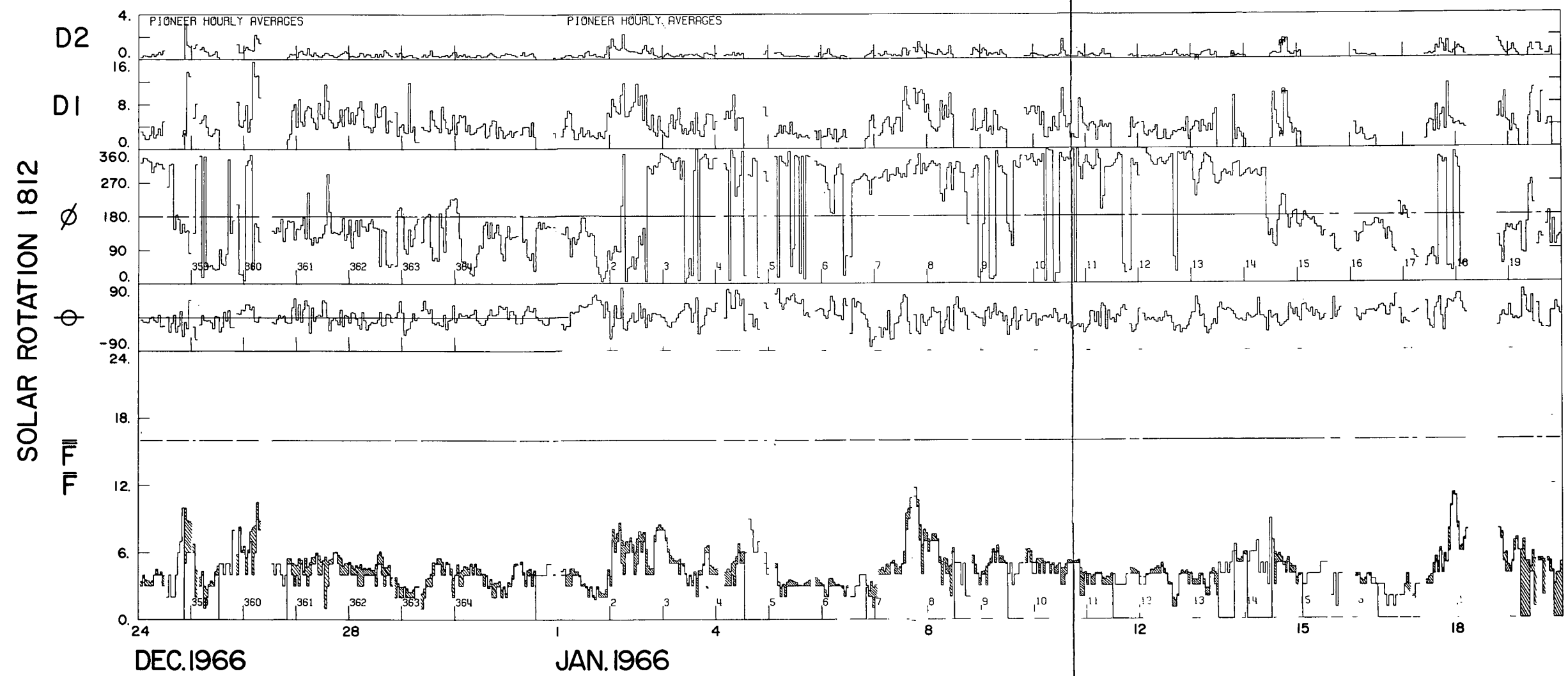


FIGURE 3



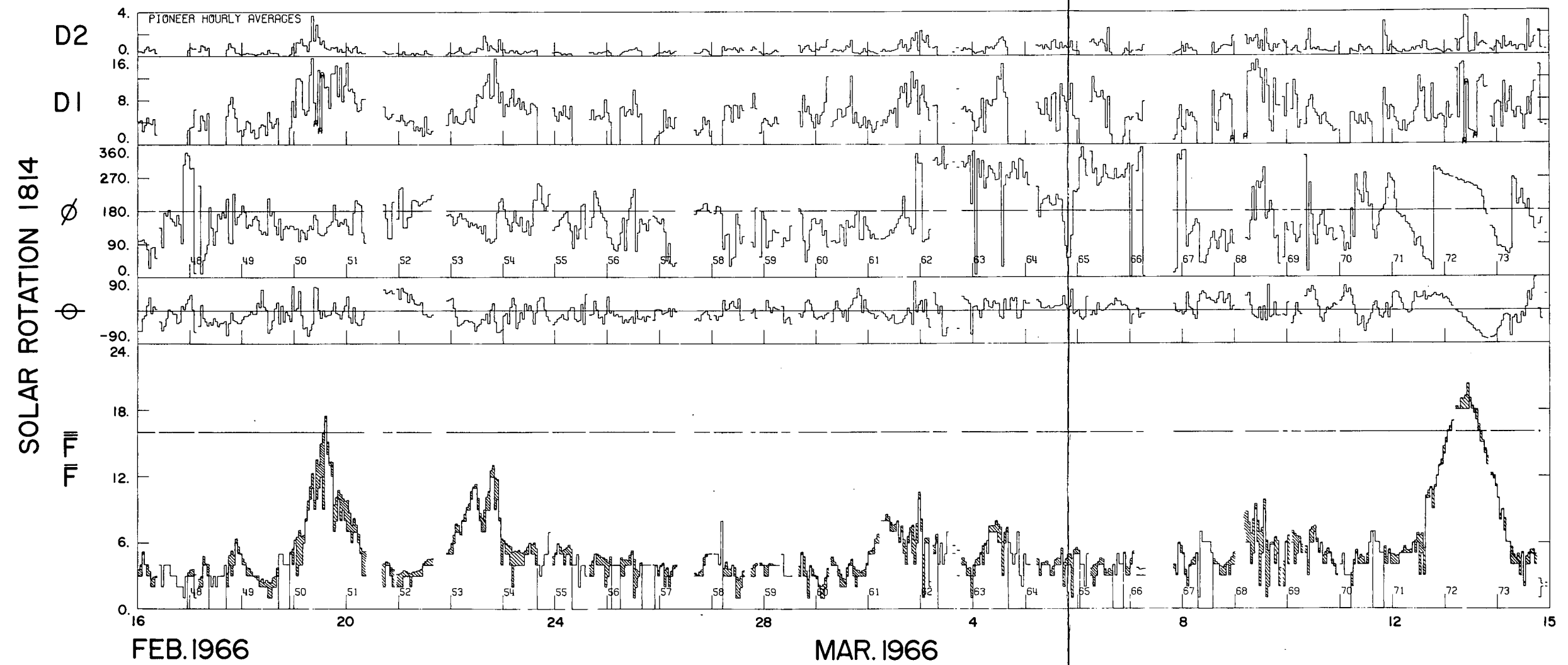


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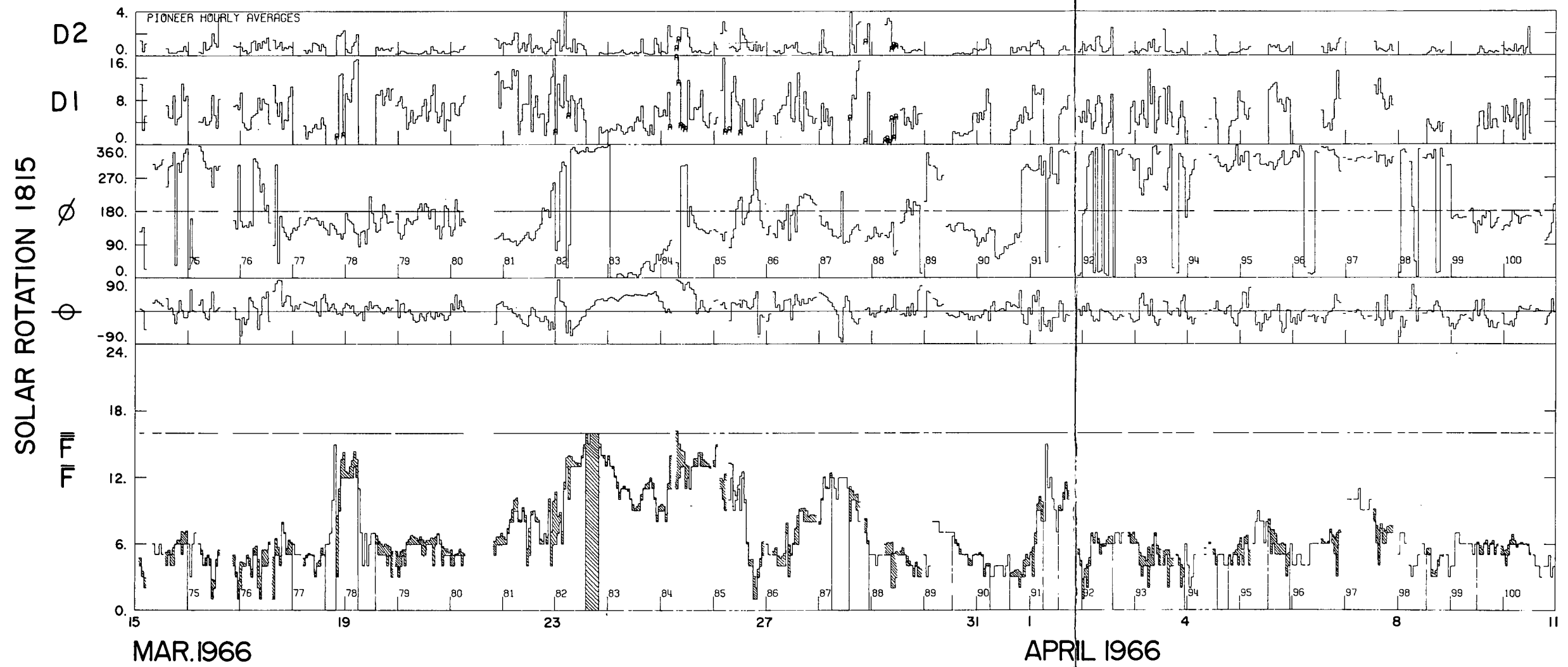


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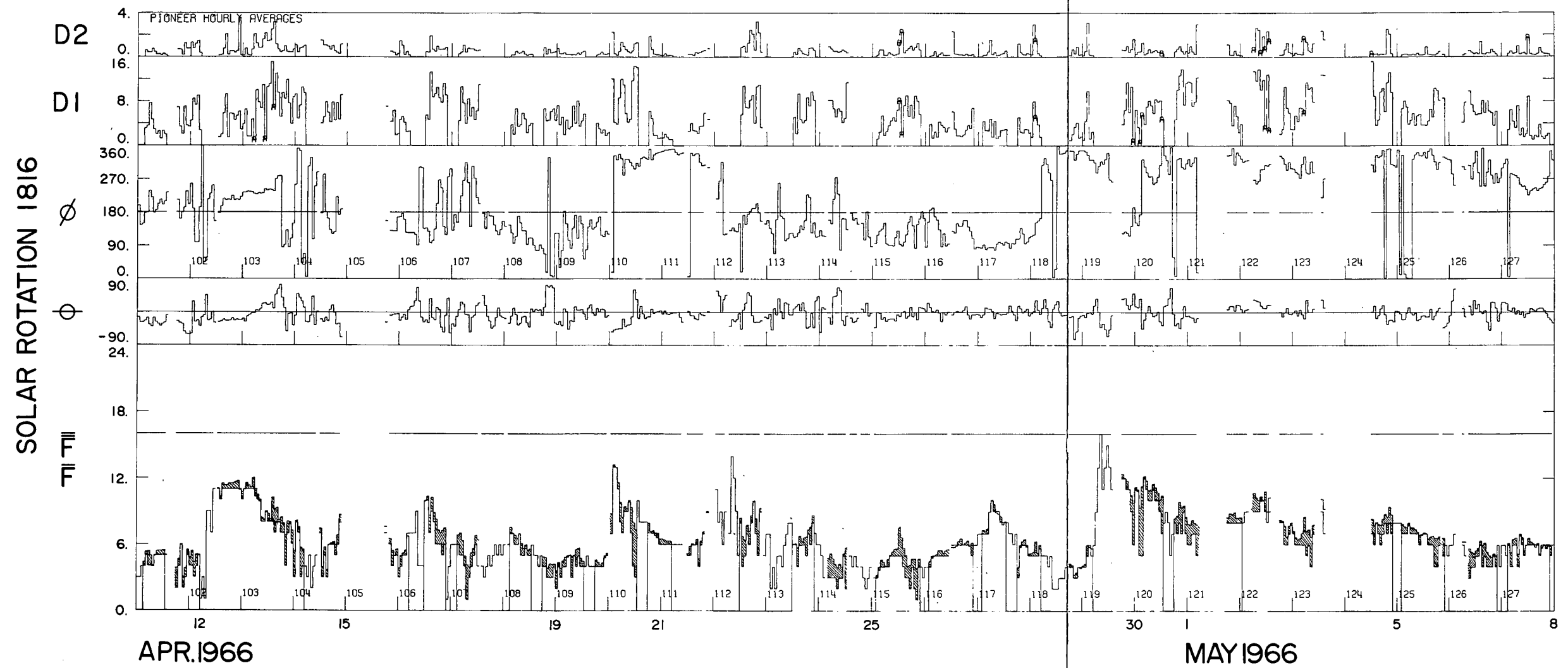


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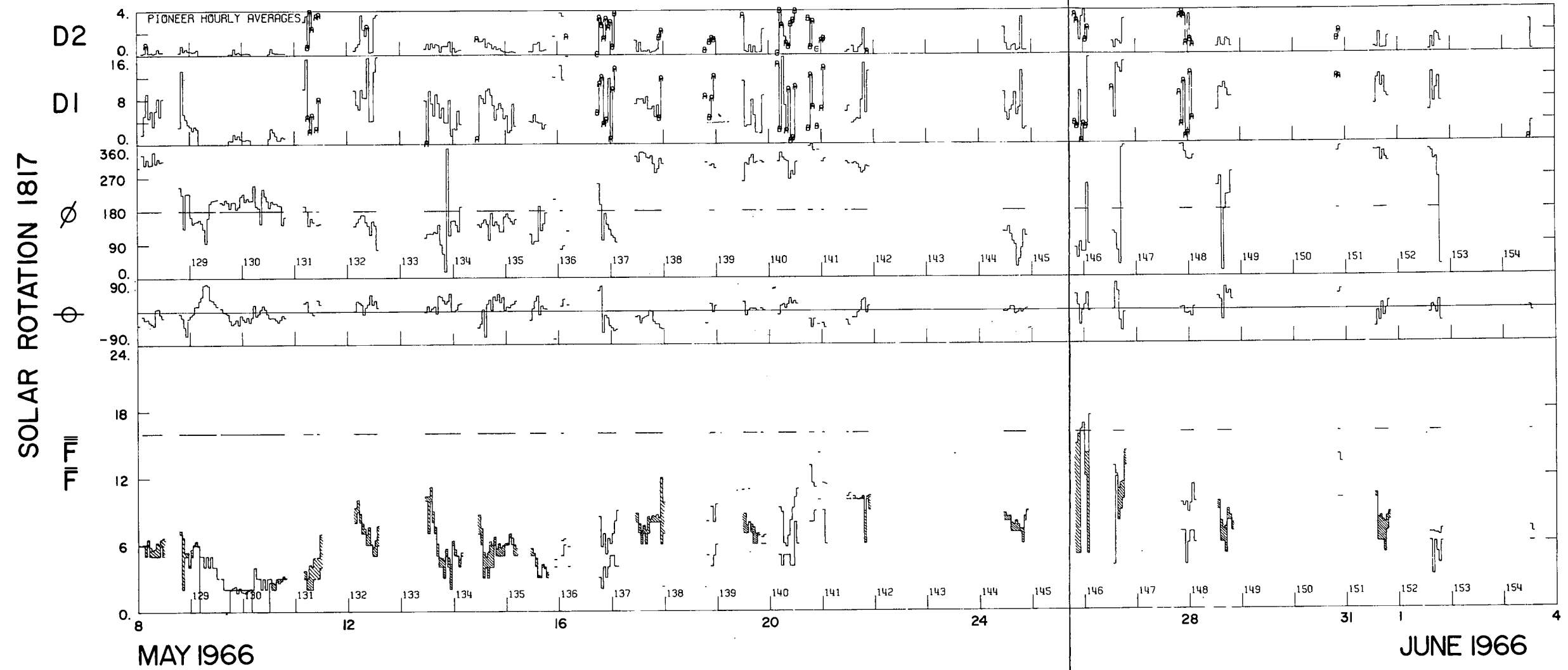


FIGURE 8



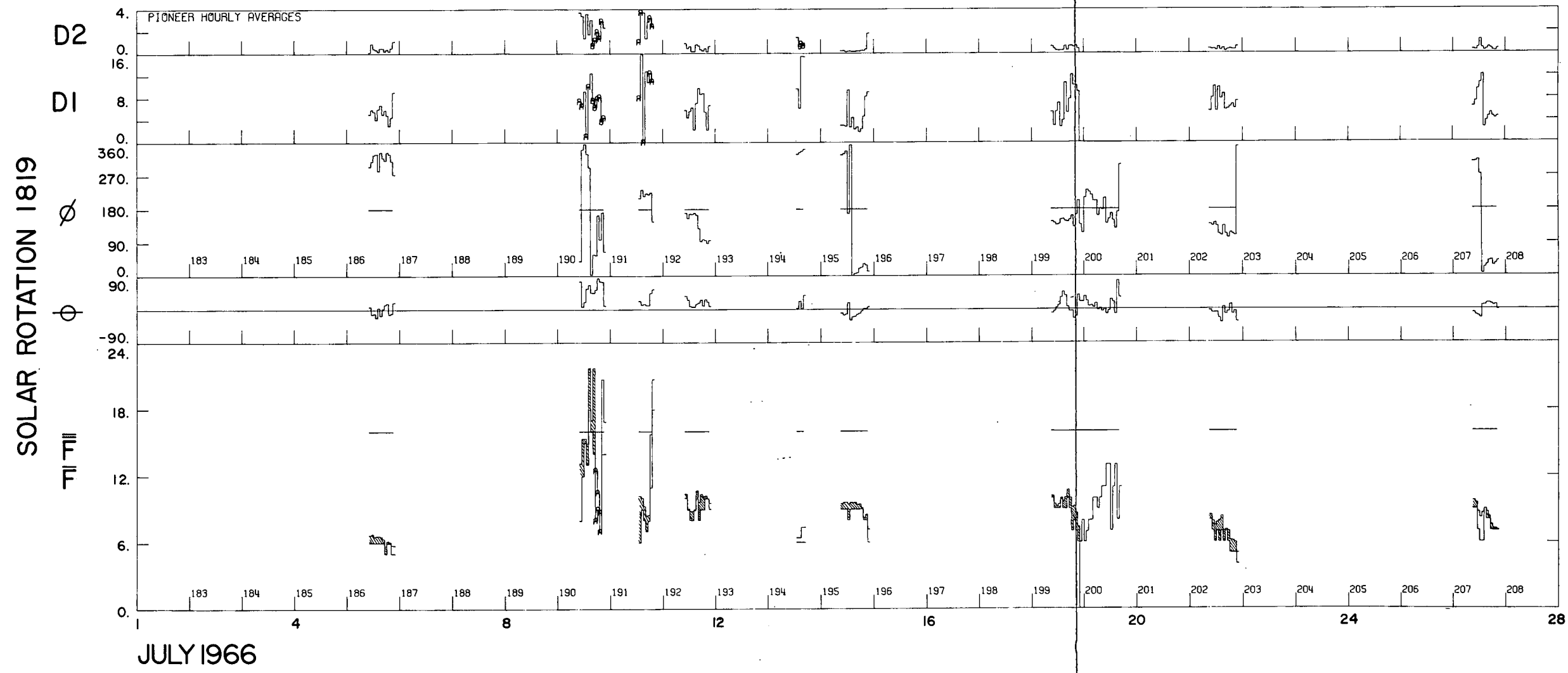


FIGURE 10

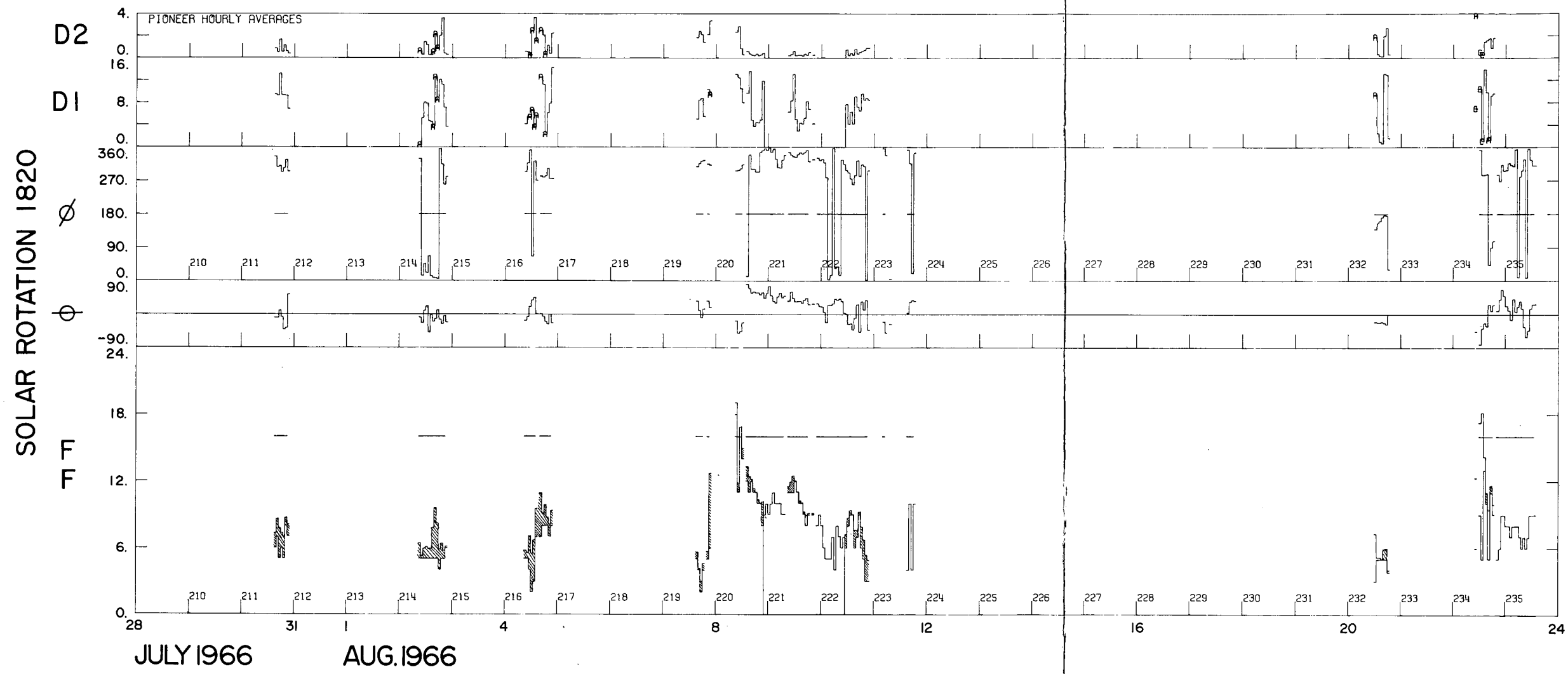


FIGURE 11

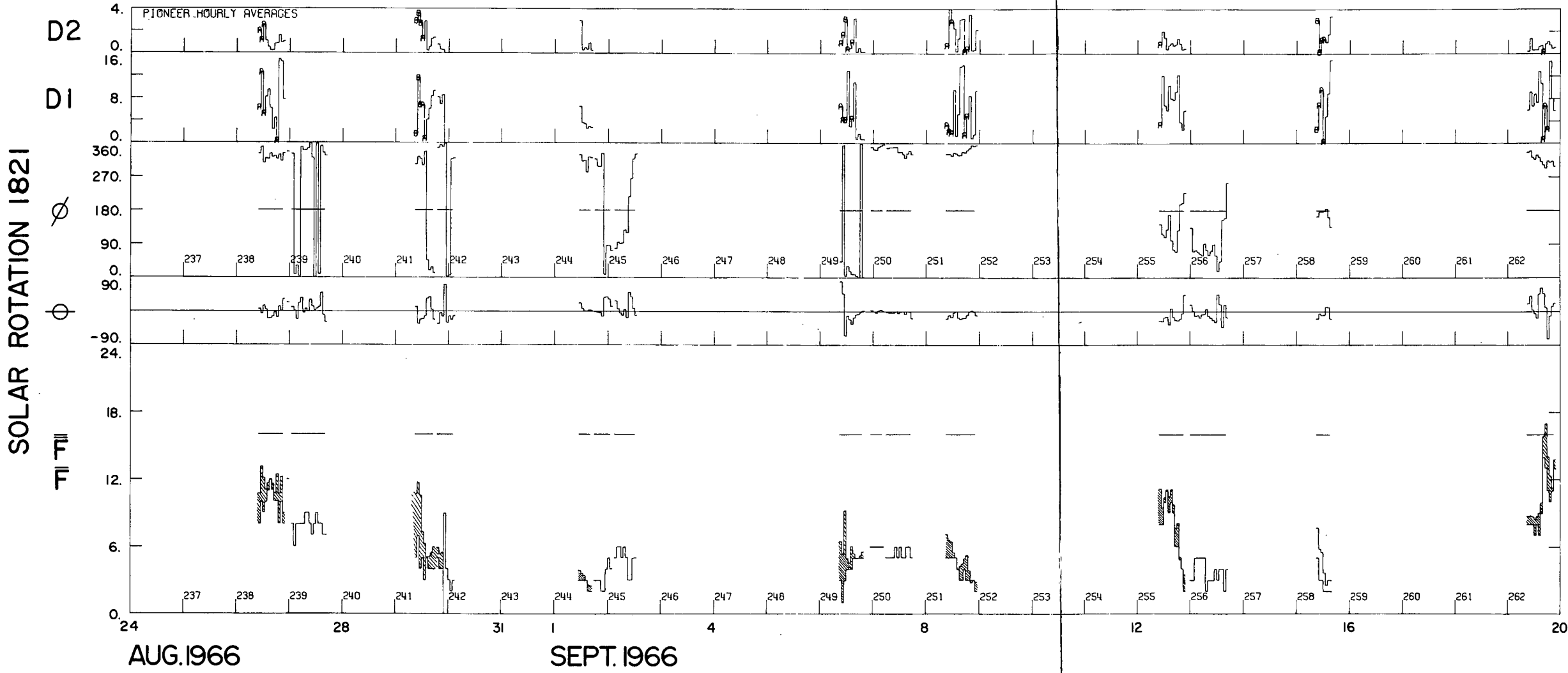


FIGURE 12

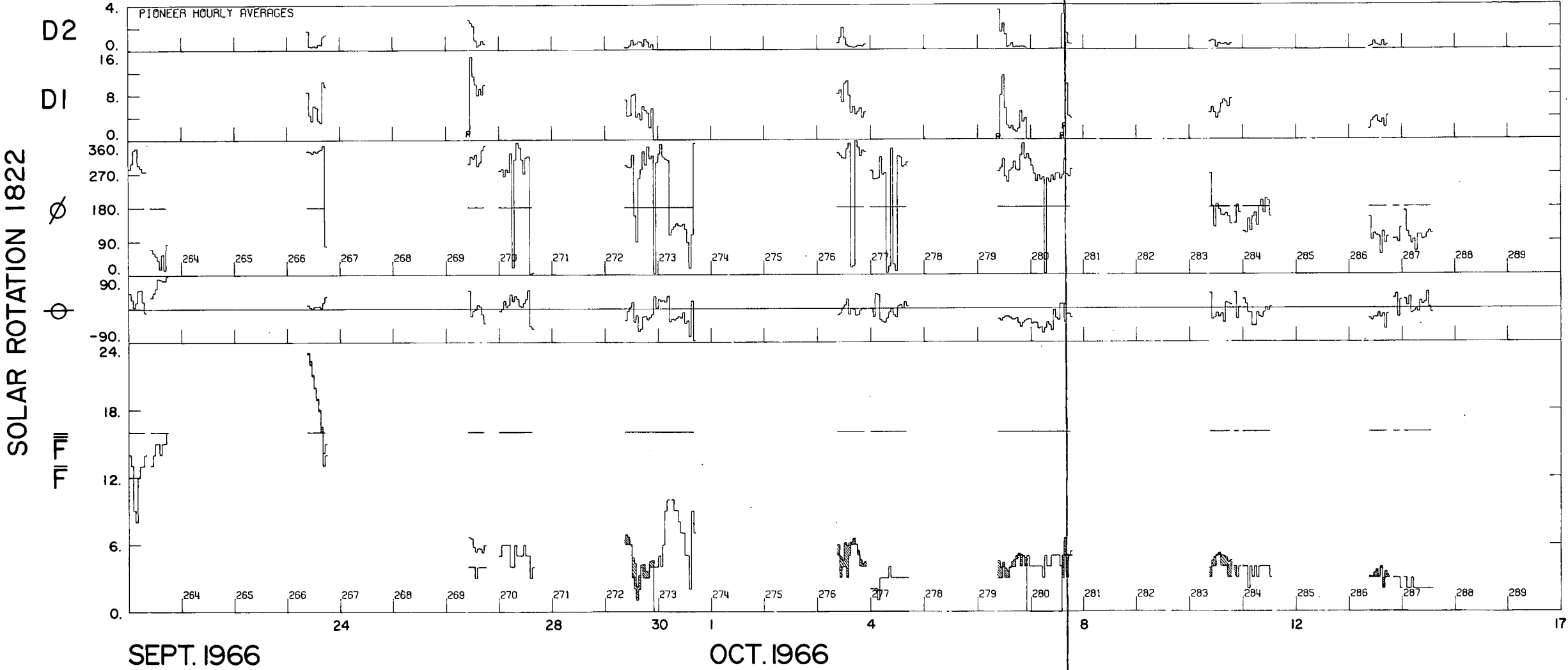


FIGURE 13

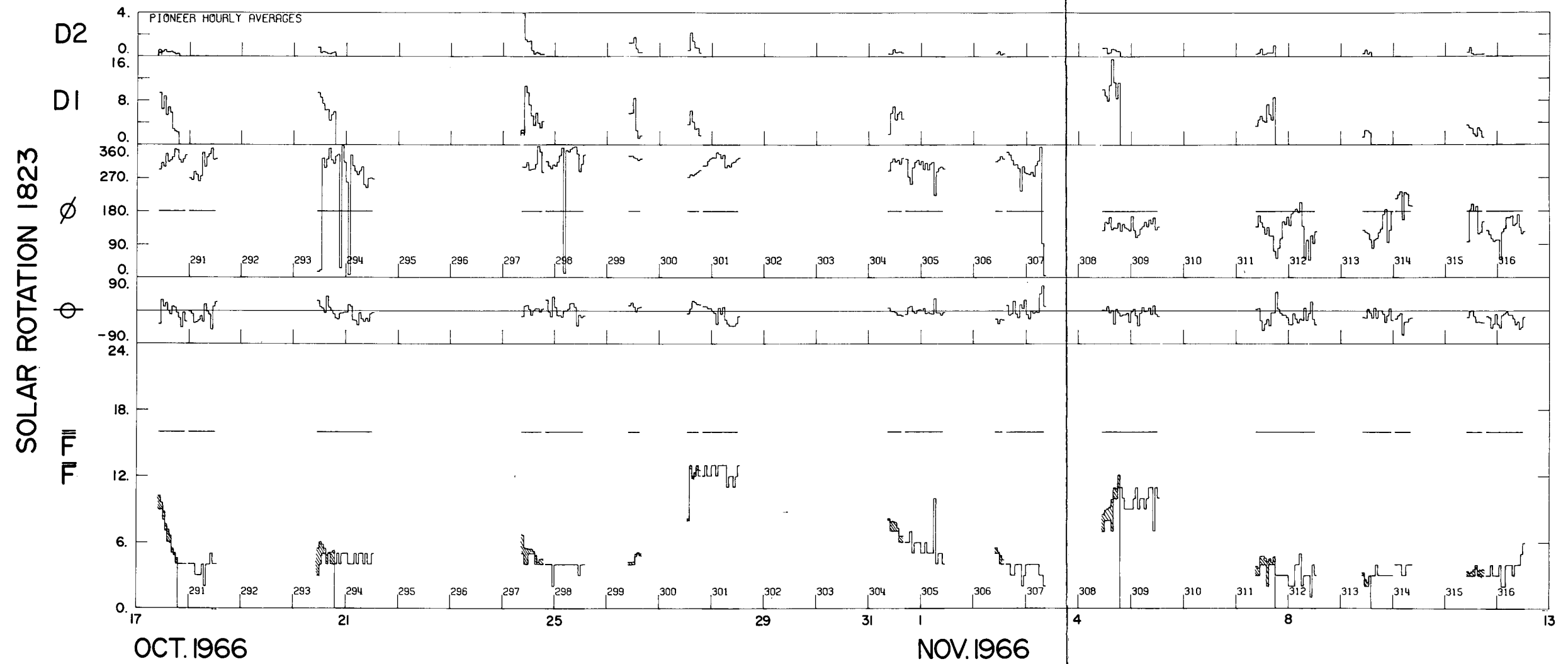


FIGURE 14

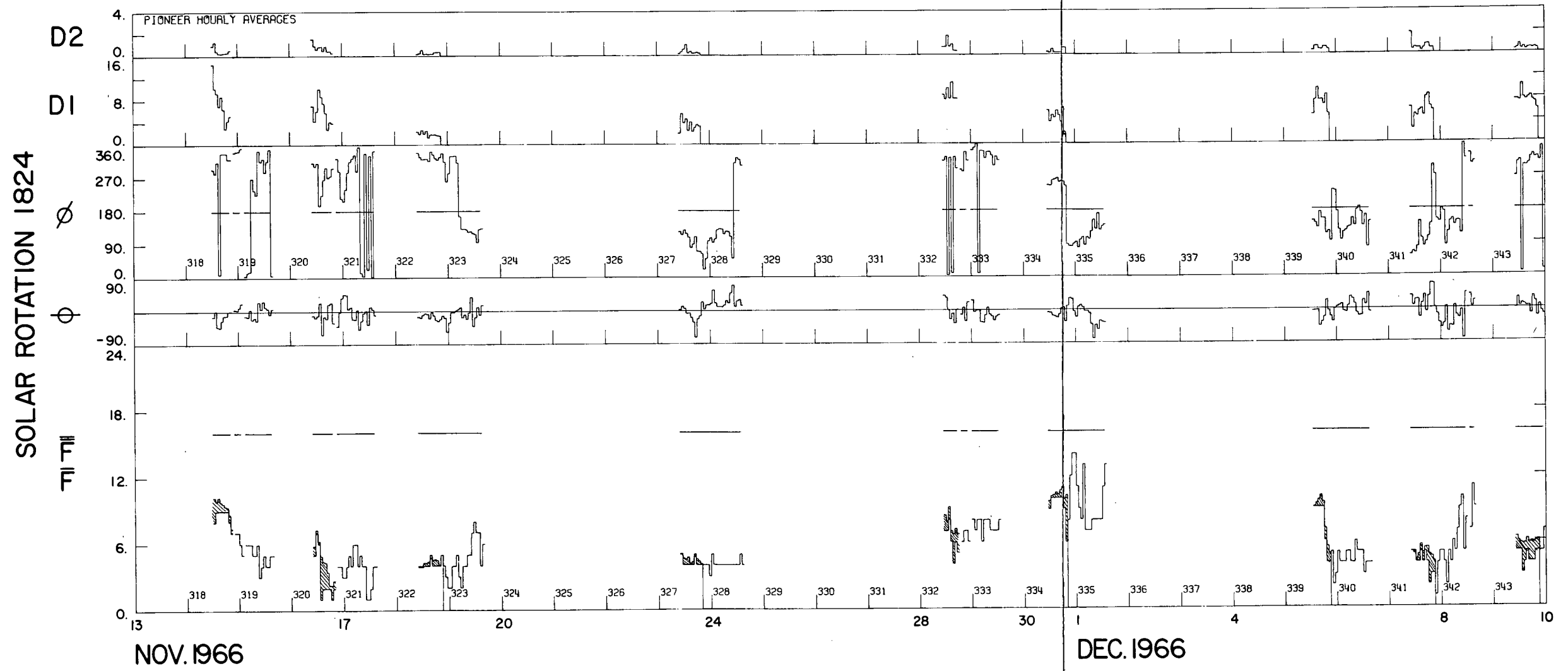


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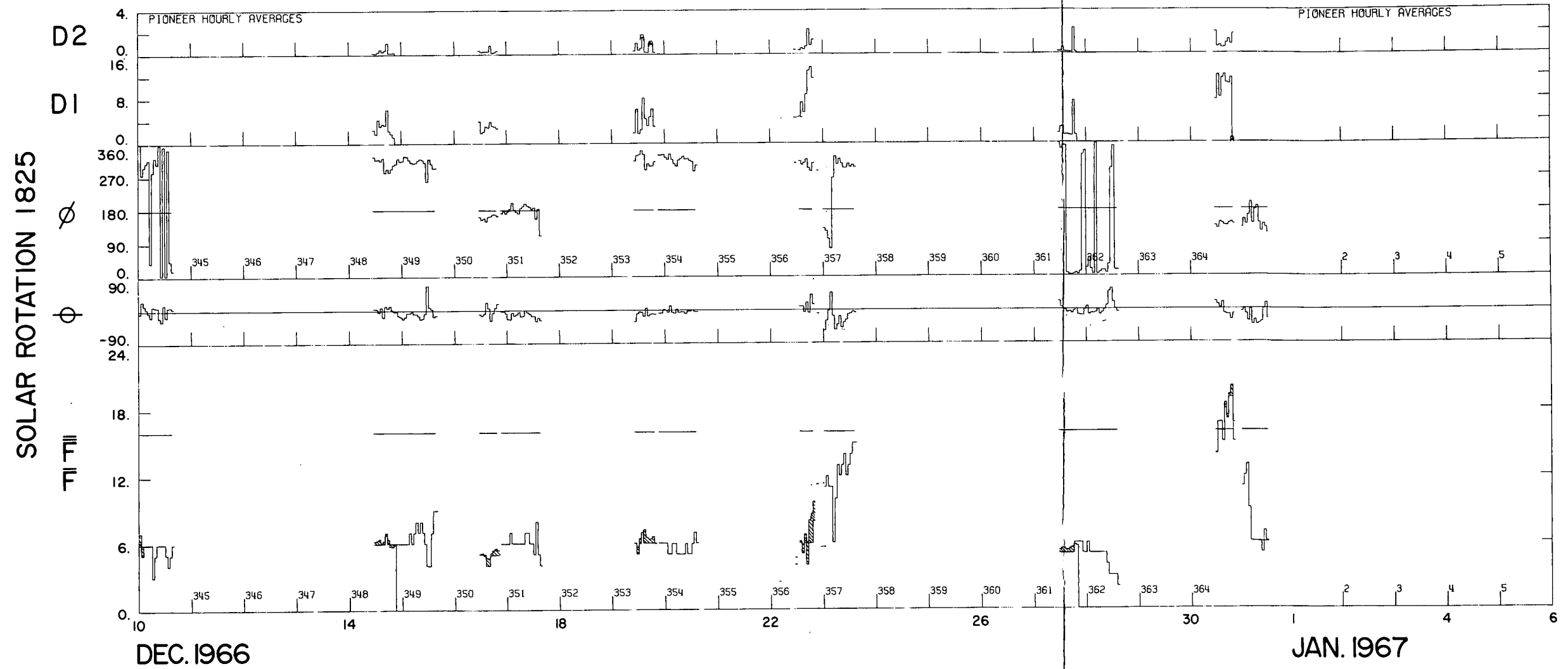


FIGURE 16

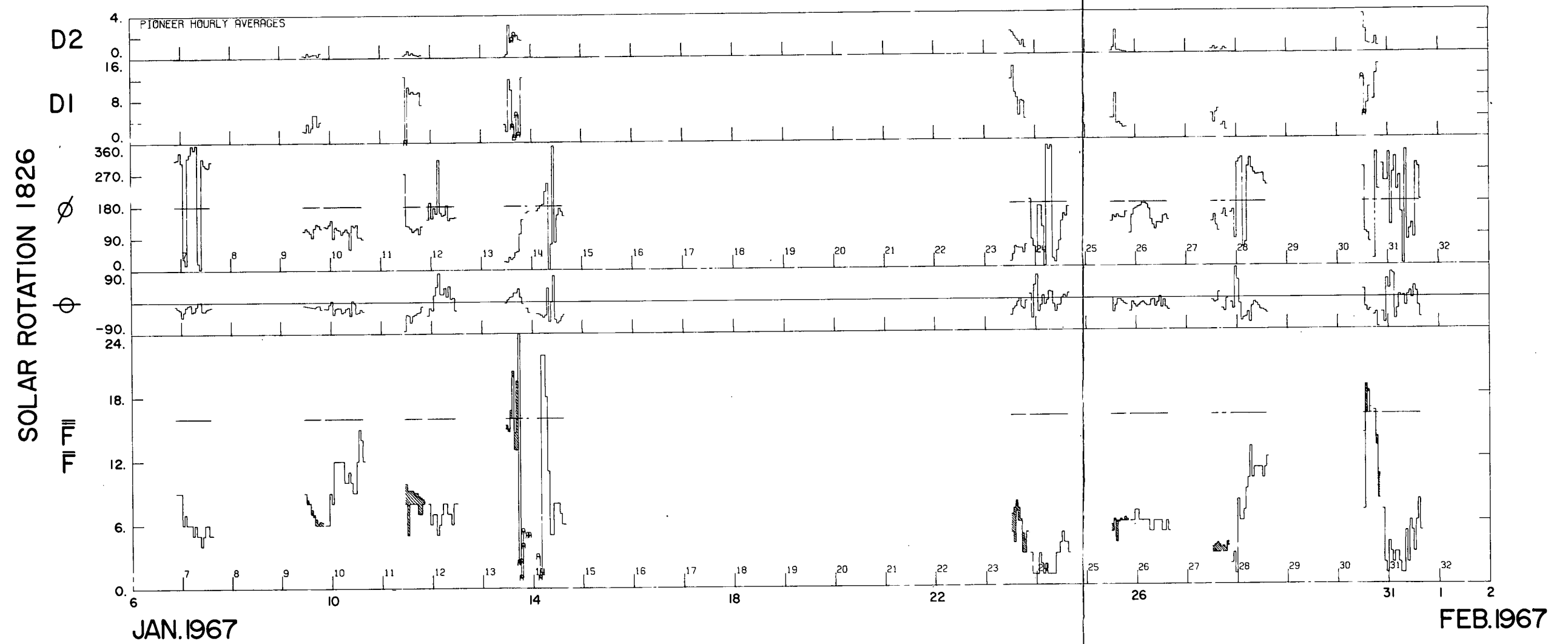


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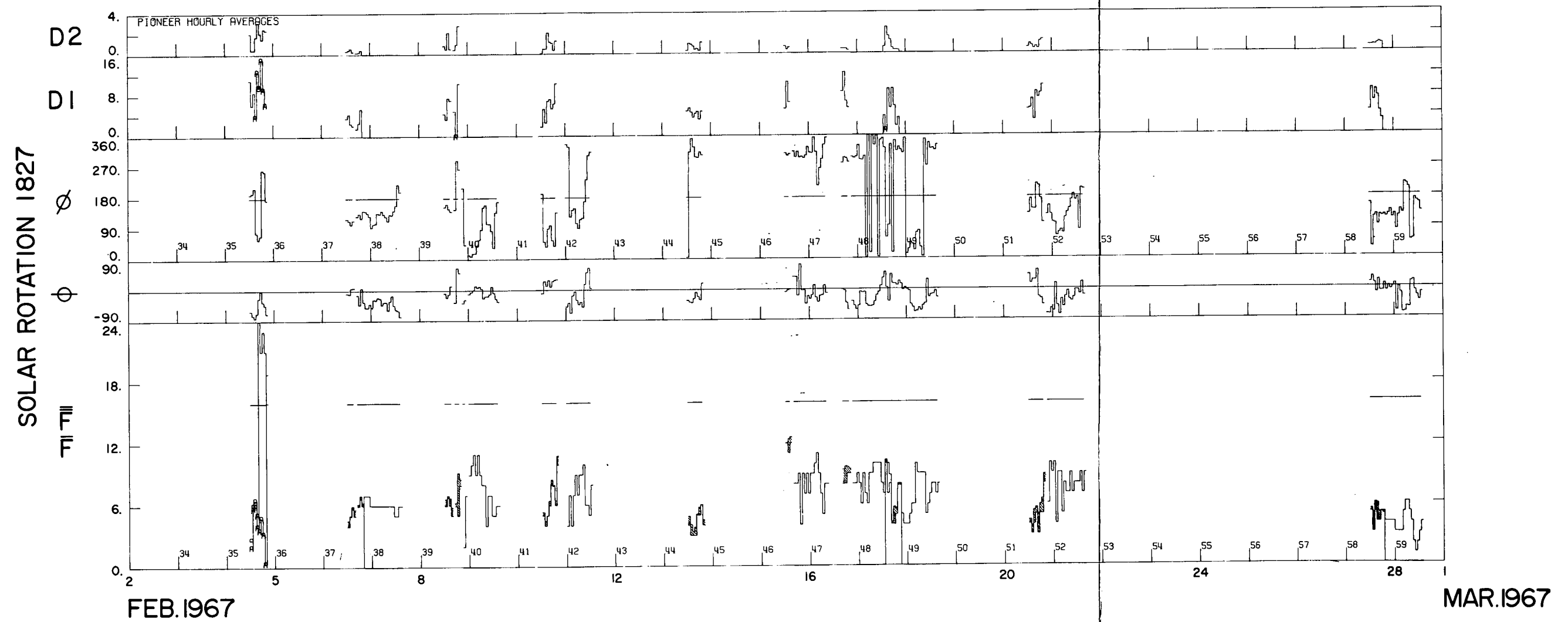


FIGURE 18

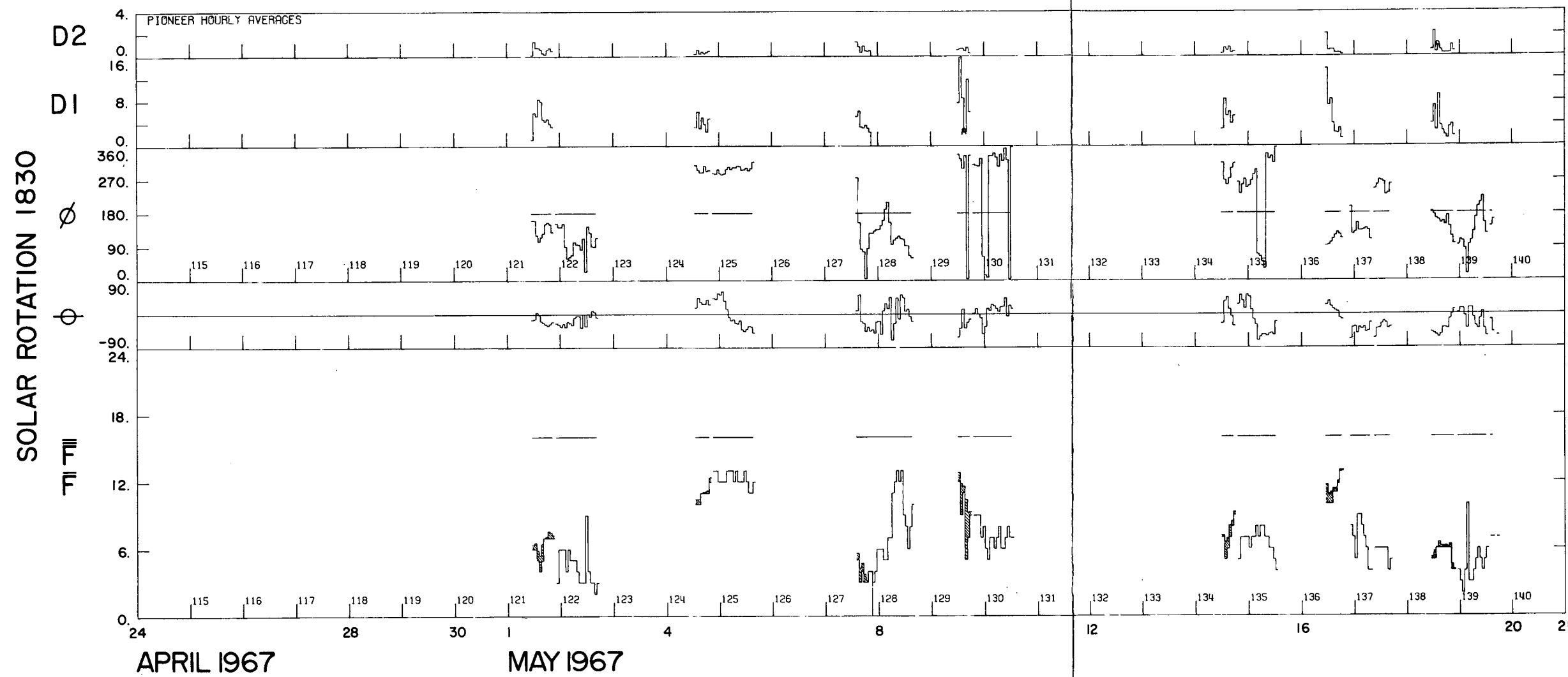


FIGURE 19

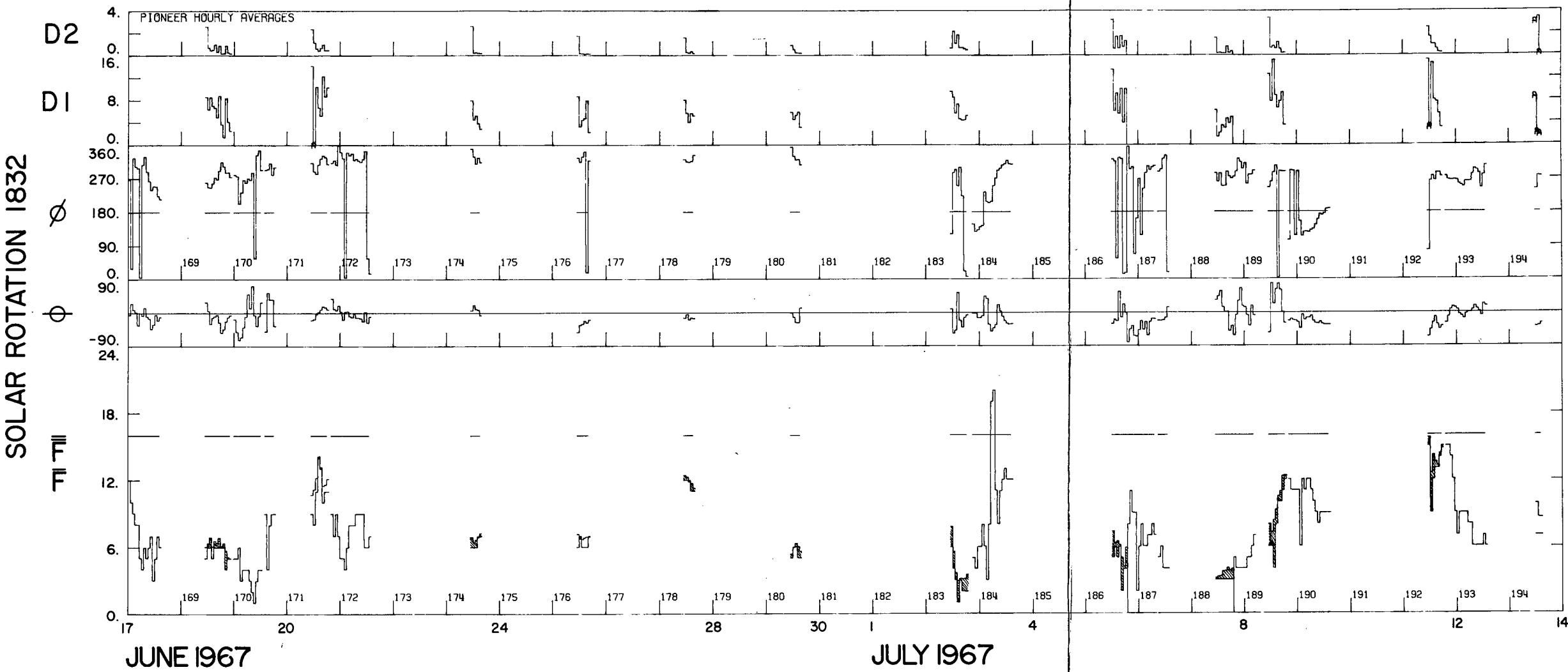


FIGURE 20

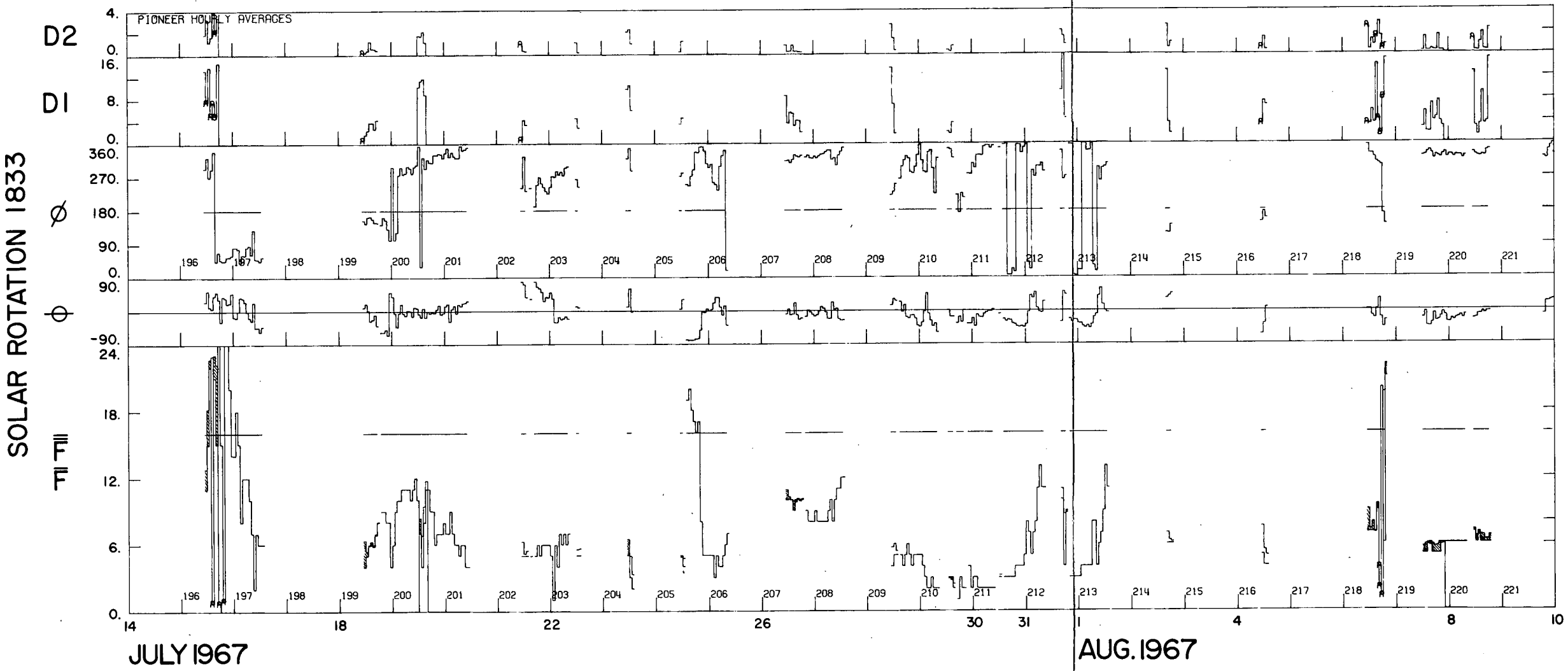


FIGURE 21

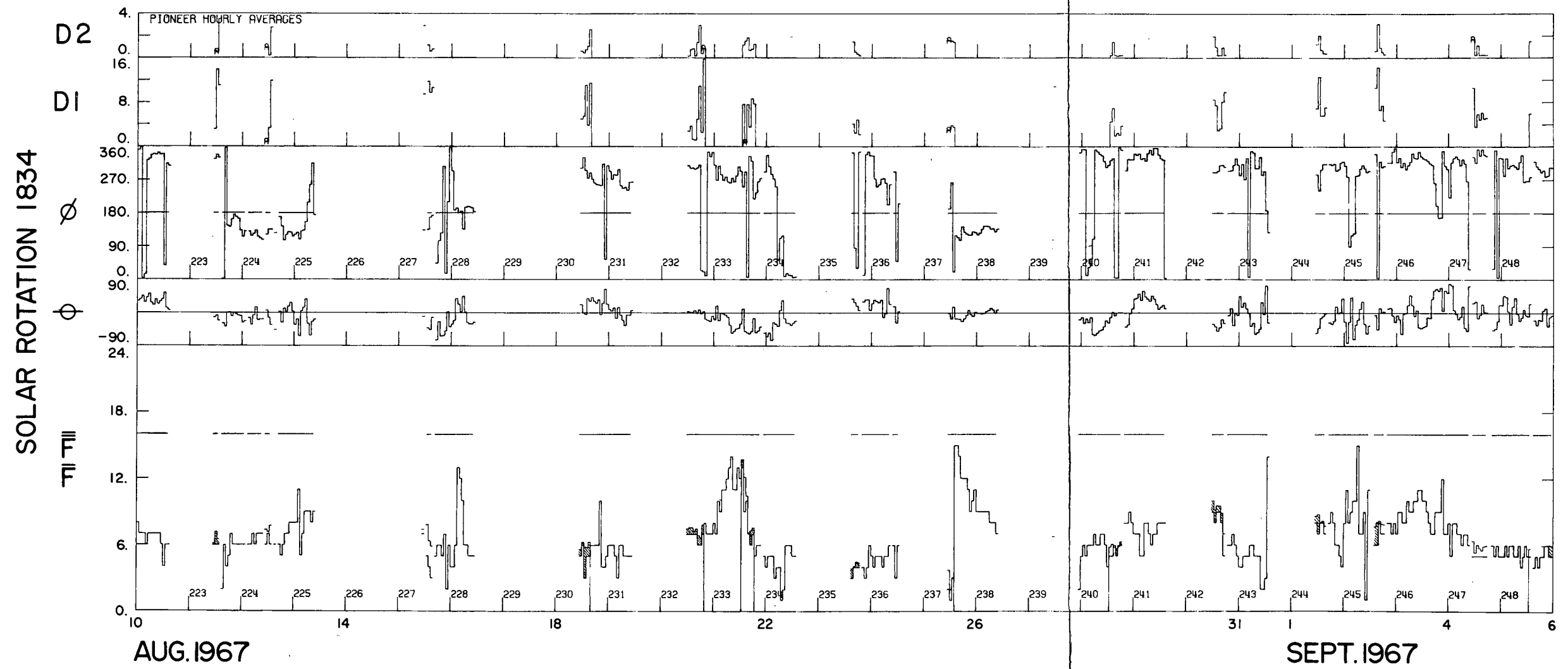


FIGURE 22